Effect of Thermal Stresses on the Thermal Expansion and Damping Behavior of ZA-27/Aluminite Metal Matrix Composites

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When the fabrication of a metal matrix composite (MMC) involves its cooling from a high temperature, plastic-elastic residual deformation fields can be generated within and around the particle due to the differential thermal expansion between the particle and matrix metal. The present investigation is concerned with the effect of thermal residual stresses on the thermal expansion and damping behavior of aluminite particulate-reinforced ZA-27 alloy MMCs. Composites were prepared by the compocasting technique with 1, 2, 3, and 4 wt.% of aluminite reinforcement. Thermal expansion and damping properties have been studied experimentally as a function of temperature over a temperature range 30 to 300 °C both in the **heating and cooling cycle. The thermal expansion studies exhibited some residual strain, which increased with the increase in the weight percent of the reinforcement. The damping capacity of both the composites and matrix alloy is found to increase with the increase in temperature during the heating cycle, whereas in the cooling cycle, damping behavior exhibits a maximum, which becomes more pronounced with the increase in the weight percentage of the reinforcement. The appearance of the maximum may be linked with dislocation generation and motion as a result of plastic deformation of the matrix at the metal/ reinforcement interface. This phenomenon is attributed to the thermal stresses generated as a result of coefficient of thermal expansion (CTE) mismatch between the composite constituent phases. The thermal stresses have been estimated in both the cases using simple models.**

In recent years, a great deal of progress has been achieved is still lacking.
in the development and production of particulate-reinforced $\frac{Z_{\text{inc-all}}}{Z_{\text{inc}} + \frac{1}{2}}$ in the development and production of particulate-reinforced Zinc-aluminum (ZA) alloys are a relatively a new family
metal matrix composites (MMCs). When compared with unremetal matrix composites (MMCs). When compared with unre-
inforced metals/alloys, MMCs exhibit significant improvements
characteristics [10] Their fabrication requires lower energy in strength, elastic modulus, $[1,2]$ wear resistance, $[3]$ fatigue resis-
tance, $[4]$ and damping capacity $[5]$ in addition to high-temperature
allows have a wide range of freezing temperature resulting in tance,^[4] and damping capacity^[5] in addition to high-temperature
mechanical properties^[6] and low thermal expansion.^[7] A low
coefficient of thermal expansion (CTE) and high damping
composition.^[11] A mong the thermal residual stresses on the properties of materials. When

MMCs are fabricated at a certain high temperature and cooled

down to room temperature, residual stresses are induced into the

matrix and reinforcement becau

Keywords aluminite, CTE, damping capacity, thermal uents. The residual stresses would have some undesirable effects stresses, ZA-27 on the mechanical, thermal, and physical properties of the composites. A number of articles on experimental work and numeri-**1. Introduction 1. Introduction 1. Introduction 1.** Introduction **1.** Interature.^[8,9] However, due to the complexity of the MMCs, a better understanding of the residual stresses in these materials

 $characteristics^[10]$ Their fabrication requires lower energy

ated in the composite as a result of the differences in the CTEs between the matrix and reinforcement have been evaluated Shanta Sastry, Department of Physics, NMKRV College for Women,
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and **Jayagopal Uchil,** Department of Material Science, Mangalore University, Mangalore, India. \blacksquare at which they are brought into contact. At any other temperature,

Aluminum	Cu	Mg	Zn
$25 - 28%$	$1 - 2\%$	$0.01 - 0.02\%$	Balance

there will be biaxial or triaxial stress fields due to the difference
in the physical and mechanical properties of the constituents
for the composite. The thermal stresses may be of considerable
of the composite. The therm ies of MMCs must exhibit these residual strains and at the **2.3 Damping Measurement** same time dislocation movements should contribute to MMCs damping behavior.
A dynamic mechanical analyzer (DMA, model 983, DuPont)

compocasting technique was used to prepare the composite,
which is similar to the one used by Sharma et al.^[18] In this were obtained. process, the matrix alloy (ZA-27) was first superheated above its melting temperature and stirring was initiated to homogenize the temperature. The temperature was then lowered gradually **3. Results** until the alloy reached a semisolid state. At this temperature (440 8C), the preheated aluminite particles were introduced into **3.1 Microstructure** molten slurry. Stirring was continued until the interface between
the particles and the matrix promoted wetting. The melt was
then superheated above its liquidus temperature of 500 °C and
finally poured into the lower die

Specimens for CTE testing were machined from the prepared MMC samples with the dimensions $10 \times 5 \times 5$ mm. The specimen surfaces were ground with series of silicon carbide

Table 1 Chemical composition of ZA-27 alloy (ASTM papers of 100, 200, 400, 600, and 1000 grit and then polished **B669-82**) with 3 μ m diamond paste to obtain a fine-surface finish. The specimens were then washed in distilled water, followed by acetone, and then allowed to dry thoroughly. About four specimens of each sample were tested to achieve the reproducibility of experimental results. Percent linear change (PLC) measurements were performed from 30 to 300 $^{\circ}$ C at the rate of 5 $^{\circ}$ C/ min with commercial thermal mechanical analyzer equipment

includes sample arms and clamps, flexure pivot, LVDT, electro-**2. Experimental Procedure 2. Experimental Procedure** a computer. The specimen is subjected to a flexural sinusoidal **2.** Computer. The specimen is subjected to a flexural sinusoidal strain with constant amplitude, and resultant bending stress is **2.1 Material Preparation** measured simultaneously. The sample is held between two end In the present study, the ZA-27 alloy having the chemical
composition as per the ASTM B669-82 ingot specifications
given in Table 1 has been used as the base alloy. The reinforce-
ment is basically a hydrous sulfate havin $g/cm³$ and particle size of about 100 μ m. Its chemical composi-
tion is Al₂O₃, SO₃, 9H₂ O. A thermogravimetric analysis (TGA)
study of the aluminite sample confirms the fact that at higher
temperatures it

low Cu content in the ZA-27 will lead to formation of an **2.2 CTE Measurement**

Specimens for CTE testing were machined from the prepared

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eutectic reaction given by^[21]

$$
L \to \alpha + \eta + \varepsilon \tag{Eq 1}
$$

At a lower temperature of $268 \degree C$, a ternary phase known as Al₄Cu₃Zn (*T'*) will result from the reaction^[20]

$$
\varepsilon + \alpha \to T' + \eta \tag{Eq 2}
$$

The microstructure of the ZA-27 as-cast alloy consisted of a cored aluminum-rich matrix (α, fcc) , and interdendritic zincrich phase (η , hcp), CuZn₄ (ε), and Al₄Cu₃Zn (T'). The light etching constituent shown in Fig. 1(a) is the aluminum-rich (α) phase, while the grey structure surrounding it is a mixture of zinc-rich (η) and aluminum-rich phases. The dark regions are zinc-rich (η) phase.

The microstructures (Fig. 1b) of the aluminite-reinforced composites show negligible solid solubility of aluminite in either the aluminum- or zinc-rich phases. The grain size of the ZA-27 matrix alloy is somewhat larger than that of the composites. As a result, there were large clusters of aluminite within some areas of the matrix, while other areas were entirely aluminite depleted. The segregation was more pronounced in the 4% aluminite composites. When analyzed at higher magnifications (Fig. 1c), the structures reveal that the relatively colder particle chilled the metal and initiated nucleation. The dendrites grow away from the particle, due to the restriction caused by the particle-to-solute enrichment. Thus, the grains grow outward from the particle and the last remaining eutectic liquid solidifies around the particles. However, no gap is observed between the particle and the matrix, and the particles are well bonded with the matrix.

3.2 Coefficient of Thermal Expansion

The results of CTE expressed as a PLC of different weight percentages of particle-reinforced MMCs as a function of temperature are shown in Fig. 2. These curves exhibit some residual strain on cooling, which increases with the increase in the weight percentage of reinforcement. The residual strain in the case of base alloy has the minimum value. The variation of CTE of the composites as well as the base alloy with temperature is shown in Fig. 3. The curves show that the CTE decreases with the increase in the weight percentage of the reinforcement. But in all the cases there is a moderate increase in CTE values with the increase in temperature.

3.3 Damping Capacity

The damping capacity (tan ϕ) versus temperature curves for the base alloy and composites both in the heating and cooling cycles are shown in Fig. 4. The damping capacity of the composites as well as matrix alloy has been found to increase with the **Fig. 1** Microstructure of (a) ZA-27 matrix alloy and ZA-27/4% alum-
increase in temperature and also with the increase in reinforce-
inite MMCs at (b) lower ment. But the heating and cooling curves exhibit hysteresis behavior in all cases, which becomes more prominent with the increase in the weight percent of the reinforcement. In the heating cycles of all the cases, a maxima (peak) is observed at 300 °C, while featuring isotropic reinforcements such as aluminite particulate-
in the cooling cycles, the in the cooling cycles, the peak appears at about 225 $^{\circ}$ C.

ties of matrix and reinforcement. Comparatively, composites

inite MMCs at (**b**) lower magnification and (**c**) higher magnification

chanical behavior. The PLC versus temperature curves in the **4. Discussion**
 4. Discussion
 4. Discussion
 4. Discussion deforming plastically at a slightly lower temperature and exhib-Though CTE is a basic physical property, it is relatively ited a relatively larger residual contraction than the unreinforced difficult to predict the same in the case of MMCs because it alloy. In order to calculate these thermal stresses, an expression is influenced by several factors, which include physical proper-
ties of matrix and reinforcement. Comparatively, composites and Friedberg^[22] has been used, which is given by

Fig. 2 Graphs showing PLC vs temperature of ZA-27/aluminite **MMCs**

$$
P = \frac{(\alpha_d - \alpha_m) \Delta T}{[(1 + \mu_m)/2E_m] + [(1 - 2\mu_d)/E_d]} \qquad \text{(Eq 3)} \qquad \text{interaction} \qquad \text{maximum.}
$$

interface; α_m , μ_m , and E_m represent the CTE, Poisson's ratio, and elastic modulus of the matrix, respectively; and α_d , μ_d , for activation energy corresponding to the thermally activated and E_d represent the same quantities for the reinforcement. relaxation process is given b and E_d represent the same quantities for the reinforcement. Substituting $\alpha_m = 26 \times 10^{-6}$ °C, $\alpha_d = 7.4 \times 10^{-6}$ °C, $\mu_m =$ 0.33, $\mu_d = 0.22$, $E_m = 79$ GPa, and $E_d = 275$ GPa, Eq 3 gives

$$
P = 1.778 \text{ MPa} \times \Delta T \qquad (\text{Eq 4})
$$

For a change in temperature of $1 \degree C$, the thermal stresses are

Thermal stresses in MMCs have already been studied by internal friction or damping capacity measurements.^[24] The existence of the damping maximum is particularly important since this maximum could be related to the thermal stresses, which are generated in the composites. In the absence of thermal where ν^* is the activation volume and is given by $\nu^* = 400$ composite specimen is cooled, tensile thermal stresses arise in by thermal expansion studies.

Fig. 3 The graph showing CTE vs temperature of ZA-27/aluminite MMCs

the matrix due to the particulate distribution and induce a longrange movement of the dislocations, which is superposed on the oscillatory motion imposed by the damping apparatus. This interaction is supposed to be the origin of the observed maximum.

The vibration frequency and strain amplitude dependence where *P* is the pressure or stress at the matrix/reinforcement of the damping maximum suggests that the relaxation processes interface; α_m , μ_m , and E_m represent the CTE, Poisson's ratio, are of thermal origin and

$$
f\tau_0 \exp\left(E/kT_P\right) = 1/2\pi \qquad \qquad \text{(Eq 5)}
$$

where *f* is the vibration frequency, τ_0 is the relaxation time, *E* is the activation energy, k is the Boltzmann constant, and T_p is the peak temperature. The data used in evaluating activation calculated to be 1.778 MPa at the interface between the matrix energies E_h and E_c , corresponding to heating and cooling peaks, and reinforcement, which is of the same order obtained by respectively, of the 4% aluminite-reinforced composite are other studies.^[23] given in Table 2.
The thermal stresses generated in the composite are given by
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The thermal stresses generated in the composite are given by

$$
\sigma_{th} = (E_h - E_c)/2\nu^* \tag{Eq 6}
$$

stresses, the dislocation can vibrate around its equilibrium posi- \mathbf{b}^3 (\mathbf{b} is the length of the Burgers vector of dislocations in the tion and contribute to the damping. This partly accounts for matrix alloy = 0.25 nm). A value of $\sigma_{th} = 1.359$ MPa was the observed damping capacity of the base alloy. When the calculated, which is in close agreement wit calculated, which is in close agreement with the value obtained

Fig. 4 Graph showing damping capacity vs temperature of ZA-27/ aluminite MMCs

energies 50, pp. 13-22.

Parameters	Heating peak	Cooling peak
Peak temperature (T_p) in °C (K)	300 (573)	225 (498)
Relaxation time (τ_0) in μ s	2.5427	8.0574
Vibration frequency (f) in Hz	10	10
Activation energy (calculated), eV	0.4323	0.3261

In PLC and damping tests, the hysteresis observed between
heating and cooling curves is attributed to internal thermal
thermal thermal 1986, vol. 17A, pp. 379-89.
1986, vol. 17A, pp. 379-89. stresses existing in the composites due to CTE mismatch 17. H.M. Ledbetter and M.W. Austin: *Mater. Sci. Eng.,* 1987, vol. 89, pp. between matrix and reinforcement during fabrication and subse-

quent thermal cycling. Theses thermal stresses change sign 18. S.C. Sharma, B.M. Girish, D.R. Somashekar, R. Kamath, and B.M. quent thermal cycling. Theses thermal stresses change sign
during heating and cooling (compressive stresses during heating
and tensile stresses during cooling).^[26] The hysteresis is also and tensile stresses during cool with constituent elements having different properties, which *Sci. Lett.*, 1996, vol. 15, p. 1008.
may lead to mismatch stresses between them. This accounts 21. Yuanyuan Li, Tungwai Leo Nag for the small hysteresis in the matrix alloy, which increases a 1996, vol. 198, pp. 126.
with the increase in the reinforcement in the composites.
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pp. 5104-07.

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has been studied over a temperature range 30 to 300 °C both in the heating and cooling cycles. Thermal expansion studies showed residual strains and maxima were obtained in the heating and cooling curves of the damping behavior of the composite. These investigations revealed the presence of residual thermal stresses generated in the composite due to the difference in the CTE between the matrix and reinforcement. The longrange mobility of dislocations in the metal matrix around the reinforcement seems to be the critical parameter for stress relaxation at the matrix/reinforcement interface. This parameter has been characterized by damping measurements. The thermal expansion study of thermal stresses leading to plastic deformation in the matrix and residual strain obtained is particularly useful in any application of the composite at elevated temperatures. The thermal stresses have been evaluated in both cases of thermal and damping studies and found to be in good agreement with each other.

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